Abstract
A generalized approach was developed for solving contact problems in a dynamic elastic-plastic formulation. For the design of composite and reinforced materials, a technique for solving dynamic contact problems in more adequate an elastic-plastic mathematical formulation is used. To consider the physical nonlinearity of the deformation process, the method of successive approximations is used, which makes it possible to reduce the nonlinear problem to a solution of the sequences of linear problems. In contrast to the traditional plane strain, when one normal stress is equal to a certain constant value, for a more accurate description of the deformation of the sample, taking into account the possible increase in longitudinal elongation, we present this normal stress as a function that depends on the parameters that describe the bending of a prismatic body that is in a plain strain state. The problem of a plane strain state of a beam made from the composite reinforced one-layer material is being solved. The reinforced or armed composite material consists of two materials: the main material of glass and two rows of the reinforcing crystalline fourteen fibres of basalt. Glass has high strength and is not affected by the processes of aging of the material, corrosion, and creep. In addition, this material is cheap and widely available. The reinforced composite beam is rigidly linked to an absolutely solid base and on which an absolutely solid impactor acts from above in the centre on a different size of the area of initial contact.

Keywords: Plane, strain, stress, state, impact, composite, armed, reinforced, material, elastic-plastic, deformation.

Introduction
The use of a generalized approach to solving dynamic contact problems in an elastic-plastic formulation makes it possible to use it to solve contact problems for a body of arbitrary shape, which is subjected to an arbitrary distributed over the contact zone or shock loading.

Since glass is a cheap, ubiquitous material that is not susceptible to corrosion and aging and creep processes, like metals and alloys, the study of composite materials containing glass is relevant and actual. Glass is also convenient in that it can be poured into the frame of the reinforcement and thus can be further strengthened. As reinforcing elements, metal wire, polysilicate, polymer, polycarbon, crystalline compounds, which can have a fairly small thickness, can be used.

In (Bogdanov, 2023; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022), a new approach to solving the problems of impact and nonstationary interaction in the elastoplastic mathematical formulation was developed. In these papers like in non-stationary problems (Bogdanov, 2023; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022), the action of the striker is replaced by a distributed load in the contact area, which changes according to a linear law. The contact area remains constant.

The solution of problems for composite cylindrical shells (Lokteva et al., 2020), elastic half-space (Igumnov et al., 2013), elastic layer (Kuznetsova et al., 2013), elastic rod (Fedotenkov et al., 2019; Vahterova & Fedotenkov, 2020) were developed using method of the influence functions (Gorshkov & Tarlakovsky, 1985).

In (Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022, 2022) dynamic interaction process of plane hard body and two layers reinforced composite material was investigated and the fields of summary plastic deformations and normal stresses arising in the base are calculated using plane strain (Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022) and plane stress (Bogdanov, 2022) states models. In (Bogdanov, 2022) results are depending on the size of the area of an initial contact between the impactor and the upper surface of the base. In (Bogdanov, 2022) results were
calculated depending on the thickness of top metal layer of the composite base. In (Bogdanov, 2022) results were calculated depending on the material of top layer of the composite base. It was investigated composite bases reinforced by steel, titanium and aluminium top layers.

In contrast from the work (Bogdanov, 2018), in this paper, we investigate the impact process of hard body with plane area of its surface on the top of the composite beam which consists main glass layer reinforced by seven crystalline basalt fibers.

**Problem Formulation**

Deformations and their increments (Bogdanov, 2023), Odquist parameter \( \kappa = \int d\varepsilon_i^p \) (\( \varepsilon_i^p \) is plastic deformations intensity), stresses are obtained from the numerical solution of the dynamic elastic-plastic interaction problem of infinite composite beam \([-L/2 \leq x \leq L/2, -\infty \leq y \leq \infty, -h \leq z \leq h] \), in the plane of its cross section in the form of rectangle. It is assumed that the stress-strain state in each cross section of the beam is the same, close to the plane deformation, and therefore it is necessary to solve the equation for only one section in the form of a rectangle \( \Sigma = L \times B \) with two materials: main glass layer \([-L/2 \leq x \leq L/2, -\infty \leq z \leq \infty] \), and two rows of fourteen reinforcing crystalline basalt fibres \( [|x| \leq b_1; -\infty \leq z \leq \infty; h_1 - h \leq y \leq h_1] \), \( [bi \leq |x| \leq bi+1; -\infty \leq z \leq \infty; h_i - h \leq y \leq h_i] \), \( |x| \leq b_i; -\infty \leq z \leq \infty; B - h_i - h \leq y \leq h_i] \), \( [bi \leq |x| \leq bi+1; B - h_i - h \leq B - h_i] \) (i=2; 4; 6). The contact between glass and basalt fibres is ideally rigid. We assume that the contact between the lower surface of the reinforced glass base and the absolute hard half-space \( \{y \leq 0\} \) is ideally rigid.

From above on a body the absolutely rigid drummer contacting along a segment \( \{|x| \leq A; y = B\} \). Its action is replaced by an even distributed stress -\( P \) in the contact region, which changes over time as a linear function \( P = p_0(t) + p_0(t)^2 \). Given the symmetry of the deformation process relative to the line \( x=0 \), only the right part of the cross section is considered below (Fig. 1). The calculations use known methods for studying the symmetry of the deformation process relative to the line \( x=0 \). The strain state in each cross section of the beam is the same, close to the plane deformation, and therefore it is necessary to solve the equation for only one section in the form of a rectangle. The equations of the plane dynamic theory are considered, for which the components of the displacement vector \( \mathbf{u} = (u_x, u_y) \) are related to the components of the strain tensor by Cauchy relations:

\[
\varepsilon_{xx} = \frac{\partial u_x}{\partial x}, \quad \varepsilon_{yy} = \frac{\partial u_y}{\partial y}, \quad \varepsilon_{xy} = \frac{\partial u_x}{\partial y} + \frac{\partial u_y}{\partial x},
\]

The equations of motion of the medium have the form:

\[
\frac{\partial \sigma_{xx}}{\partial x} + \frac{\partial \sigma_{xy}}{\partial y} = \rho \frac{\partial^2 u_x}{\partial t^2}, \quad \frac{\partial \sigma_{xy}}{\partial x} + \frac{\partial \sigma_{yy}}{\partial y} = \rho \frac{\partial^2 u_y}{\partial t^2}, \tag{1}
\]

where \( \rho \)– material density. The boundary and initial conditions of the problem have the form:

\[
x = 0, \ 0 < y < B: \ u_x = 0, \ \sigma_{xy} = 0, \n\]

\[
x = L/2, \ 0 < y < B: \ \sigma_{xx} = 0, \ \sigma_{xy} = 0, \n\]

\[
y = 0, \ 0 < x < L/2: \ u_y = 0, \ \sigma_{xy} = 0, \n\]

\[
y = B, \ 0 < x < L/2: \ \sigma_{xx} = 0, \ \sigma_{xy} = 0. \n\]

\[
u_{ik}|_{y=0} = 0, \ \nu_{sik}|_{y=0} = 0, \ \dot{u}_k|_{t=0} = 0, \ \ddot{u}_s|_{t=0} = 0. \quad (3)
\]

The determinant relations of the mechanical model are based on the theory of non-isothermal plastic flow of the medium with hardening under the condition of Huber-Mises fluidity. The effects of creep and thermal expansion are neglected. Then, considering the components of the strain tensor by the sum of elastic-plastic behaviour under the condition of Huber-Mises fluidity. The effects of creep and thermal expansion are neglected. Then, considering the components of the strain tensor by the sum of elastic-plastic behaviour under the condition of Huber-Mises fluidity.

![Fig. 1: Geometric scheme of the problem](image-url)
statement requires further research.

The material is strengthened with a hardening factor $\eta^*$ (Bogdanov, 2023; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Mahnenko, 1976):

$$\sigma_s(T) = \sigma_{02}(T_0) \left[ 1 + \frac{K(T)}{E_0} \right]^\eta^*, \quad E_0 = \frac{\sigma_{02}(T_0)}{E},$$

(6)

where $T$ – temperature; $K$ – Odquist parameter, $T_0 = 20^\circ C$, $\eta^*$ – hardening coefficient; $\sigma_s(T)$ – yield strength after hardening of the material at temperature $T$. Rewrite (4) in expanded form:

$$d\sigma_{xx} = \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{xx} + \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{yy},$$

$$d\sigma_{yy} = \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{yy} + \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{xx},$$

(7)

In contrast to the traditional plane deformation, when $\Delta E_{xx} (x, y) = \Delta \epsilon_{xx}^0 + \Delta \epsilon_{xx}^1 x + \Delta \epsilon_{xy}^1 y$, where unknown $\Delta \epsilon_{xx}^1$ and $\Delta \epsilon_{xy}^1$ describe the bending of the prismatic body (which simulates the plane strain state in the solid mechanics) in the Oxz and Oyz planes, respectively, and $\Delta \epsilon_{xx}^0$ – the increments according to the detected deformation bending along the fibres $x = y = 0$.

**Solution Algorithm**

The nonstationary interaction (Bogdanov, 2023) occur in a time interval $t \in [0, t_1]$. Then for every moment of time $t$:

$$e_{xx}^0 = \frac{\sigma_{xx} - \sigma}{2G} + K \sigma, \quad e_{yy}^0 = \frac{\sigma_{yy} - \sigma}{2G} + K \sigma, \quad e_{zz}^0 = \frac{\sigma_{zz} - \sigma}{2G} + K \sigma,$$

$$d\sigma_{xx}^0 = \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{xx}^0 + \left( \frac{\sigma_{02} - \sigma}{2G} + K \sigma \right) d\sigma_{yy}^0,$$

(9)

For numerical integration over time, Gregory’s quadrature formula (Bogdanov, 2023; Hemming, 1972) of order $m = 3$ with coefficients $D_k$ was used. After discretisation in time with nodes $t_k \in [0, \Delta t]$, for each value $k$ we write down the corresponding node values of deformation increments:

$$\Delta \sigma_{xx,k} = \beta_l \sigma_{xx,k} + \beta_2 \sigma_{yy,k} + \beta_3 \sigma_{zz,k} + \beta_4 \sigma_{xy,k}, \quad \Delta \sigma_{yy,k} = \beta_5 \sigma_{xx,k} + \beta_6 \sigma_{yy,k} + \beta_7 \sigma_{zz,k} + \beta_8 \sigma_{xy,k}, \quad \Delta \sigma_{zz,k} = \beta_9 \sigma_{xx,k} + \beta_{10} \sigma_{yy,k} + \beta_{11} \sigma_{zz,k} + \beta_{12} \sigma_{xy,k},$$

$$\beta_l = \frac{1}{2} (K + 2G) \Delta \epsilon_{xx,k}, \quad \beta_2 = \frac{1}{2} (K + 2G) \Delta \epsilon_{yy,k}, \quad \beta_3 = \frac{1}{2} (K + 2G) \Delta \epsilon_{zz,k}, \quad \beta_4 = \frac{1}{4} K \beta_l, \quad \beta_5 = \frac{1}{4} K \beta_2, \quad \beta_6 = \frac{1}{4} K \beta_3, \quad \beta_7 = \frac{1}{4} K \beta_4, \quad \beta_8 = \frac{1}{4} K \beta_5, \quad \beta_9 = \frac{1}{4} K \beta_6, \quad \beta_{10} = \frac{1}{4} K \beta_7, \quad \beta_{11} = \frac{1}{4} K \beta_8, \quad \beta_{12} = \frac{1}{4} K \beta_9,$$

(10)

The solution of the system (10), gives expressions for the components of the stress tensor at each step [1]:

$$\sigma_{xx,k} = \sigma_{02}(T_0) + \Delta \sigma_{xx,k}, \quad \sigma_{yy,k} = \sigma_{02}(T_0) + \Delta \sigma_{yy,k}, \quad \sigma_{zz,k} = \sigma_{02}(T_0) + \Delta \sigma_{zz,k}, \quad \sigma_{xy,k} = \sigma_{02}(T_0) + \Delta \sigma_{xy,k},$$

$$\sigma_{xx}^0 = \sigma_{yy}^0 = \sigma_{zz}^0 = \sigma_{xy}^0.$$

Function $\psi = 1/(2G + \Delta \lambda)$, which is characteristic of the yield condition, taking into account (8), (9) and (11) is:

$$\psi = \begin{cases} \frac{1}{2G} & (f < 0) \\ \frac{1}{2G} + 3\Delta \epsilon^p_{y} & (f = 0, df = 0) \\ \frac{1}{2G} & (f > 0) \quad \text{inadmissible} \end{cases}$$

(12)

$$\Delta \epsilon_{xy} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{yy} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{xx} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{xy} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{xx} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{yy} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{xx} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right), \quad \Delta \epsilon_{yy} = \frac{1}{2G} \left( \Delta \epsilon_{xx} + \Delta \epsilon_{yy} \right).$$

Considering when calculating the value $\Delta \epsilon_{xy}$, we found that its impact is so small that without reducing the accuracy of calculations can be considered $\Delta \epsilon_{xy} = 0$.

To take into account (Bogdanov, 2023) the physical nonlinearity contained in conditions (12), the method of successive approximations is used, which makes it possible to reduce a nonlinear problem to a sequence of linear problems (Bogdanov, 2023; Mahnenko, 1976; Mahnenko, 2003; Mahnenko et al., 2009):

$$\sigma^{(n)} = \sigma^{(n-1)} + \frac{1}{2G} \sigma^{(n-1)} - \sigma^{(n-1)}, \quad \text{if } \sigma_{02} < Q; \quad \sigma^{(n)} = \sigma^{(n-1)} - \frac{1}{2G} \sigma^{(n-1)} + \sigma^{(n-1)}, \quad \text{if } Q < \sigma_{02} < \sigma^I_0; \quad \sigma^{(n)} = \frac{1}{2G} \sigma^{(n-1)} - \sigma^{(n-1)}, \quad \text{if } Q > \sigma^I_0,$$

(13)

where $Q$ – the value of the largest deviation of the stress intensity $\sigma^{(n)}$ in step n from the strengthened yield strength; $n$ – is the approximation number.

Unknown (Hemming & Waner, 1964) $\Delta \epsilon_{xx}$ and $\Delta \epsilon_{yy}$ (8) are determined from the conditions of equilibrium of even with respect to x normal stresses $\sigma_{xx}$:

$$\sum_{x, y} \sigma_{xx}(x, y) dx dy = M_{xx}, \quad \rho = 1, x, y,$$

(14)

where $M_{xx} = M_{xy} = M_{yx} = 0$; where $M_{xx}$ – projection on the axis Oz of the main vector of contact stresses, and $M_{xy}$ – corresponding projections of the main moment of the forces acting on the resistance (no torsion, as noted). Given the symmetry of the problem and $\sigma_{xx}(x, y) = \sigma_{xx}(x, y)$ this equation in case of $p = x$ is satisfied automatically.

If we substitute (8) and (11) in (14), taking into account the symmetry of the integration domain with respect to x and the even of functions $\sigma_{xx,k}, \sigma_{yy,k}, \sigma_{xy,k}$, we have $\Delta \epsilon_{xx} = 0$.

A system of linear algebraic equations is obtained for the calculation of $\Delta \epsilon_{xy}^0, \Delta \epsilon_{xy}$:
The stresses and strains used above were determined for each unit cell from the numerical solution at each point in time \( t_k = k\Delta t \).

**Numerical Solution**

For both problems the explicit scheme of the finite difference method was used with a variable partitioning step along the axes Ox (M elements) and Oy (N elements). The step between the split points was the smallest in the area of the layers contact and at the boundaries of the computational domain. Since the interaction process is fleeting, this did not affect the accuracy in the first thin layer, areas near the boundaries, and the adequacy of the contact interaction modelling.

The use of finite differences (Hemming, 1972) with variable partition step for wave equations is justified in (Zukina, 2004), and the accuracy of calculations with an error of no more than \( O((\Delta x)^2 + (\Delta y)^2 + (\Delta t)^2) \) where \( \Delta x, \Delta y \) and \( \Delta t \) – increments of variables: spatial \( x \) and \( y \) and time \( t \). A low rate of change in the size of the steps of the partition mesh was ensured. The time step was constant.

The resolving system of linear algebraic equations with a banded symmetric matrix was solved by the Gauss method according to the Cholesky scheme.

In (Weisbrod & Rittel, 2000), during experiments, compact samples were destroyed in 21 – 23 ms. The process of destruction of compact specimens from a material of size and with contact loading as in (Weisbrod & Rittel, 2000) was modelled in a dynamic elastoplastic formulation as plane strain state, considering the unloading of the material and the growth of a crack according to the local criterion of brittle fracture. The samples were destroyed in 23 ms. This confirms the correctness and adequacy of the developed formulation and model.

Figures. 2 – 19 show the results of calculations of one layer specimens with a hardening factor of the material \( \eta^* = 0.05 \). The main has made from quartz glass. The reinforcing fibres have made from basalt. Contact between glass and basalt is an ideal. Calculations were made at the following parameter values: temperature \( T = 50^\circ\text{C} \); \( L = 20\text{mm} \); \( B = 5\text{mm} \); \( h = 0.1\text{mm} \); \( \Delta t = 3.21 \times 10^{-10} \text{s} \); \( P_{01} = 8 \text{MPa} \); \( P_{02} = 10 \text{MPa} \); \( M = 94 \); \( N = 103 \). The smallest splitting step was 0.02 mm, and the largest 0.82 mm (only the first layer); \( \Delta y_{\text{max}} = 0.82 \text{mm} \); \( \Delta y_{\text{max}} = 0.05 \text{mm} \); \( b_1 = h/2 \); \( b_2 = 1.05 \text{mm} \); \( b_4 = 2.15 \text{mm} \); \( b_6 = 3.25 \text{mm} \); \( b_i = b_{i-1} + h \), \( i = 3, 5, 7 \).

At the Figures. 2 – 10 and 11 – 19 show results of numerical solution of problems when the contact zone was equal \( a = 2a = a_1 = 8\text{mm} \) and \( a = a_2 = 3\text{mm} \), respectively.
Fig. 10: Stress $\sigma_{yy}$ when $a = a_1$ and $t = t_3$

Fig. 11: Odquist parameter $K$ when $a = a_1$ and $t = t_1$

Fig. 12: Odquist parameter $K$ when $a = a_2$ and $t = t_2$

Fig. 13: Odquist parameter $K$ when $a = a_2$ and $t = t_3$

Fig. 14: Stress $\sigma_{xx}$ when $a = a_2$ and $t = t_1$

Fig. 15: Stress $\sigma_{xx}$ when $a = a_2$ and $t = t_2$

Fig. 16: Stress $\sigma_{xx}$ when $a = a_2$ and $t = t_3$

Fig. 17: Stress $\sigma_{yy}$ when $a = a_2$ and $t = t_1$

Fig. 18: Stress $\sigma_{yy}$ when $a = a_2$ and $t = t_2$

Fig. 19: Stress $\sigma_{yy}$ when $a = a_2$ and $t = t_3$

Figs. 2, 5, 8, 11, 14, 17; 3, 6, 9, 12, 15, 18; 4, 7, 10, 13, 16, 19 show the fields of the Odquist parameter $K$, normal stresses $\sigma_{xx}$ and $\sigma_{yy}$ at times $t_1 = 5.46 \times 10^{-6}$ s, $t_2 = 5.78 \times 10^{-6}$ s and $t_3 = 6.1 \times 10^{-6}$ s respectively.

Figs. 9 – 20 show that the highest stresses occur in the upper layer of the metal and the process of accumulation of plastic deformations is more intense there. These Figs. show areas where the normal stresses in layers are tensile. This is due to the fact that compressive stresses arise in the upper layer quickly and the contact between the layers and the contact of the lower boundary of the lower layer with an absolutely rigid base are ideally rigid.

At Fig. 2 – 4, 11 – 13 it can be seen that the greatest plastic deformations occur in a thin layer under the contact zone. This confirms the need to strengthen the glass layer with a thin layer of steel/metal on the upper surface of the base (Bogdanov, 2023; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022; Bogdanov, 2022). Fig. 5 – 10 and 14 – 19 show how stresses are concentrated in the vicinity of crystalline basalt fibres. At the moments of time $t_1$ and $t_2$, in the vicinity of basalt fibres, the absolute values of normal stresses are more than twice as high as the absolute value of stresses in other areas. Crystalline basalt fibres affect the distribution of stresses and work as concentrators of these stresses.

When the contact zone $a = a_1$, at the moment of time $t_3$ at a depth of about 1 mm from the upper surface of the base, an area arises where the absolute value of normal stresses drops sharply and reaches a value of $\approx 35$ MPa. This arises due to the fact that the process of interaction and the resulting stresses are of a wave nature. In the case of a smaller contact zone, when $a = a_2$, at the moment of time $t = t_3$, such areas appear at approximately the same depth of 1 mm from the upper surface of the base under the contact zone, however, the normal stresses $\sigma_{xx}$ and $\sigma_{yy}$ there become tensile and reach values 253 MPa and 268 MPa, respectively. This most likely corresponds to the fact that the process of destruction of the material takes place in this area.

It is of interest to investigate the process of non-stationary interaction of the considered material reinforced with thicker basalt crystalline fibres and a thin layer of steel/metal on the
Conclusions
The developed methodology of solving dynamic contact problems in an elastic-plastic dynamic mathematical formulation makes it possible to model the processes of impact, shock and non-stationary contact interaction with the elastic composite base adequately. In this work, the process of impact on a one-layer glass base reinforced by two rows of fourteen crystalline basalt fibres is adequately modelled and investigated. The fields of summary plastic deformations and normal stresses arising in the base are calculated. The rows of fourteen basalt fibres inside the glass layer redistribute the stresses and plastic deformations that occur in the composite such base. Normal stresses are concentrated in the areas of crystalline basalt fibres. The results obtained make it possible to design new composite reinforced armed materials.

References

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